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USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK VOLUME 59. QU-22B IN-FLIGHT CREW NOISE

AEROSPACE MEDICAL RESEARCH LABORATORY, WRIGHT-PATTERSON AIR FORCE BASE, OHIO

NOVEMBER 1975

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Volume 59 QU-228 IN-FLIGHT CREW NOISE

NOVEMBER 1975

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number)

The QU-22B is a USAF light utility aircraft used for infiltration surveillance. This report provides measured data defining the bioacoustic environments at flight crew locations inside this aircraft during normal flight operations. Data are reported for one location in a wide variety of physical and psychoacoustic measures: overall and band sound pressure levels, C-weighted and A-weighted sound levels, preferred speech interference level, perceived noise level, and limiting times for total daily exposure of

personnel with and without standard Air Force ear protectors. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application", AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations applications, limitations, etc.

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PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 72310418, Measurement of Noise and Vibration Environments of Air Force Operations. Col Justus F. Rose, Jr. conducted the field measurements and performed the data analysis; Capt Nick Farinacci prepared this report.

The authors acknowledge the efforts of Mr. John N. Cole who established the data analysis requirements and assisted in the preparation of this report, and Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton who assisted in the mechanics of data processing.

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INTRODUCTION

The QU-22B is a USAF light utility aircraft used for infiltration surveillance. This aircraft, which is manufactured by the Beech Aircraft Corporation, is powered by one GTSIO-520-G reciprocating engine rated at 375 hp at 3,400 rpm maximum take-off power. The engine drives a Hartzell three-blade constant-speed, slow-turning, quiet, 2.29 m diameter propeller through a 0.6667 gear reduction. The engine is manufactured by the Teledyne Continental Motors Corporation.

This volume provides measured data defining the bioacoustic environments produced inside this aircraft. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with operations of the QU-22B aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and aerospace ground equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, aerospace ground equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. *Refer to Volume 1* (reference 1) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., in-flight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published, and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of the updated index as it is generated.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; Autovon 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

IN-FLIGHT NOISE

MEASUREMENTS

All noise measurements were made on-board a standard-configured QU-22B aircraft during typical speed, altitude, and flight maneuver conditions. These levels describe the standard QU-22B environments, but may not be representative of those levels encountered if the aircraft has been configured differently (e.g., major equipment or structural changes).

Acoustic measurements were made at one flight crew location. Table 1 lists the measurement location and test conditions as numeric/alphabetic designators which are used on the data pages. The designator 1/A means measurement location 1 and test condition A.

The microphone was randomly moved external to the headgear in a region 0.2-0.3 meter from the head and the resultant samples analyzed using a 4- or 8-second integration time to obtain a power-averaged level that effectively smooths out short-duration fluctuations and best describes the exposure.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced inside the QU-22B aircraft at the specified location. This table includes the overall, 1/3 octave band, and octave band levels. From these data, C-weighted and A-weighted sound levels, maximum permissible time for one exposure per day (AFR 161-35) with and without standard Air Force ear protectors, preferred speech interference level, and perceived noise level are calculated and presented in Table 3. These measures are widely used to assess the effects of noise on personnel and their performance.

TABLE 1

MEASUREMENT LOCATION AND TEST CONDITIONS

QU-22B, Eglin AFB, 29 Jul 1971 Serial # 69-7704

LOCATION	POSITION	HEIGHT ABOVE DECK
1	Right Seat	Seated Head Level
CONDITION	DESC	CRIPTION
A	Engine start, right door ope	n.
В	Taxiing — 20" Hg. Manifold	Pressure, 1500-2000 RPM, doors closed.
C	Engine check - 23" Hg. Mar	nifold Pressure, 2100 RPM.
D	Takeoff - 37" Hg. Manifold	Pressure, 3400 RPM.
E	Initial acceleration - gear a	and flaps up, 37" Hg. Manifold Pressure, 3400 RPM.
F	Climb - 37" Hg. Manifold pr	essure, 3400 RPM, 100 KIAS, 800' /, 4-500'/min rate of climb.
G	Same as F - 3500' /, 300'/n	nin rate of climb.
Н	Climb - 35" Hg. Manifold Pr	ressure, 3400 RPM, 8.0M PA /.
I	Normal cruise - 32" Hg. Ma	nifold Pressure, 2900 RPM, 130 KIAS, 9.5M PA.
J	Orbit cruise - 29" Hg. Manis	fold Pressure, 2900 RPM, 130 KIAS, 9.5M PA.
К	Gear down descent — 25" H 2000'/min rate of descent.	g. Manifold Pressure, 2500 RPM, 150 KIAS, 9.5M PA 🥆 ,
L	Clean descent — 29" Hg. Ma descent.	anifold Pressure, 2900 RPM, 200 KIAS, 4000'/min rate of
M	140 KIAS, 31" Hg. Manifold	Pressure, 2900 RPM, 5.0M PA.

TABLE 1 (Continued)

MEASUREMENT LOCATION AND TEST CONDITIONS

QU-22B, Eglin AFB, 29 Jul 1971 Serial # 69-7704

POSITION	DESCRIPTION
N	Dog Leg to GCA pattern 31" Hg. Manifold Pressure, 3400 RPM, 140 KIAS, 4.0M PA.
P	GCA pattern, base turn — 26" Hg. Manifold Pressure, 3400 RPM, 2.3M PA, 130 KIAS.
Q	Glide path — 32" Hg. Manifold Pressure, 3400 RPM, 105 KIAS, gear down, flaps 20%.
R	GCA final approach — 27" Hg. Manifold Pressure, 3400 RPM, 110 KIAS, 1.3M PA, gear down, flaps 20%.
S	VFR overhead traffic pattern — $32''$ Hg. Manifold Pressure, 3400 RPM (bleed to $23''$ Hg. in pitchout).
T	VFR final approach — 15" Hg. Manifold Pressure, 2800 RPM, gear down, flaps 20%.

7											3.2
NOISE SOURCE/SUBJECT:	-	OPERATION:	ION:			-					0
1											1 10 JAN 75
INFLIGHT NUISE LEVELS											PAGE F1
	4,7	4/4	,	5	LOCATIC	ON/CONDITION	NITION	3	,	;	
FREQ (HZ)		•	3				:			}	
25	96	76	104	00	76	95	76	70	9	6	
31.5	103	, c	101	, c	66	66	6	8	44	9 9	
04	105	107	106	108	101	96	98	26	93	93	
50	96	9	100	9	95	66	93	95	90	91	
63	96	16	66	6	95	93	95	91	93	93	
80	76	96	16	0	100	0	101	102	104	0	
100	96	56	109	104	105	108	108	106	116	109	
671	001	501	9 6	> 0	100	0	103	100	103	9	
200	5 0	95	9. 9	0	100	0	102	86	103	0	
250	0 00	87	16	103	103	104	101	100	80	0	
315	62	81	78	9	16	0	93	93	76	91	
004	11	77	83	96	93	93	93	06	93	90	
200	19	78	80	89	90	9.0	91	90	95	90	
630	16	92	28	06	95	91	91	91	95	68	
900	2:	11	9 1	28	50	50	6.0	90	80 0	98	
1000	2.5	2 3	22	9 1	200	0 0	4 .	3 0	000	*	
1600	1 5	0 9	25	5.4	102	0 0	102	100	20	102	
2000	69	29	72	29	80	0 80	81	81	83	82	
2500	99	65	89	16	11	79	78	78	82	29	
3150	19	63	99	14	15	79	77	77	81	62	
0004	63	63	99	14	75	7.8	77	11	83	80	
5000	61	62	†9	73	14	15	75	92	80	77	
6300	9	09	62	20	72	73	73	14	62	92	
a	9	09	62	7.1	7.1	73	73	73	80	77	
10000	20	09	61	69	69	20	20	72	78	52	
v	53	56	0.0	69	90	60	2:	22	200	* :	
19000	10	10	79	60	60	2	-	2	0	2	

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

7										3.2
NOISE SOURCE/SUBJECT!	90)	OPERATION:	- Z			^) RUN 02
ATOCOATE ATOCOATE										
TNEI TOHT MOTOR I FUELO						•				TO DAN 13
						-) PAGE F2
					OCATIO	OCATION/CONDITION	NOITION			
	1/K	1/1	1/1	1/N	1/P	1/0	1/R	1/5	1/1	
FREQ										
(НZ)										
52	16		92	93	95	91	90	93	95	
31.5	100		98	86	98	66	16	96	16	
04	96	16	16	98	100	96	26	100	98	
50	46		95	91	95	95	95	93	96	
63	46		91	06	91	46	96	95	46	
80	114	00	66	0	100	16	101	100	97	
100	107	11	107	106	105	103	103	107	101	
125	102	03	100	0	101	9	96	102	98	
160	104	101	102	66	104	100	102	102	96	
200	102	03	105	96	26	95	26	66	105	
250	93		98	66	105	102	103	106	98	
315	76		95	95	93	91	95	93	91	
007	95		95	95	95	91	06	95	98	
500	91		91	95	90	89	8	91	84	
630	06		06	06	83	06	88	06	82	
900	80		88	87	98	86	82	87	83	
1000	98		82	40	893	28	81	46	62	
1250	10		83	28	81	0.0	2	81	9	
1600	82		85	81	62	7.8	80	62	2	
2000	48		84	83	82	80	80	82	7.8	
2500	94		82	81	4	11	78	80	92	
3150	83		82	29	78	11	77	78	75	
0004	94		94	81	80	11	78	80	92	
5000	81		80	13	11	75	75	11	73	
6300	81		19	11	15	73	73	22	72	
8000	81		80	78	15	72	42	74	72	
10000	62		78	92	75	71	72	73	69	
12500	4		28	11	15	20	72	72	69	
16000	62		22	92	75	20	72	72	69	

TABLE: MEASURED SOUND F	RE SSU	PRESSURE LEVEL	(08)) IDENTIFICATIONS) OMEGA 3.2
NOISE SOURCE/SUBJECT!	-	OPERATION	. NO			~		23) RUN 01
QU-228 AIRCRAFT INFLIGHT NOISE LEVELS											10 JAN 75
	1	,		1	OCATIO	LOCATION/CONDITION	NITION		:	;	
FREQ (HZ)			3		3	\$	3	5		3	
31.5	101		109	110	103	102	102	102	66	66	
63	100		103	102	101	103	102	102	105	101	
125	102		110	108	110	110	110	107	117	110	
250	95		96	104	105	105	108	102	105	105	
200	82		85	96	26	96	96	95	97	46	
1000	78		62	89	91	91	91	90	9.0	88	
2000	73		75	82	94	87	84	84	87	85	
0007	68		20	7.8	29	82	81	81	98	84	
8000	99		99	75	75	77	77	78	84	81	
16000	63	63	99	72	72	72	73	75	81	11	
OVERALL	109	110	113	113	112	113	113	110	117	113	

.

TABLE: MEASURED SOUND P 2 OCTAVE BAND	PRESSURE LEVEL (08)	LEVEL	(08)) OMEGA 3.2
NOISE SOURCE/SUBJECT:	-	OPERATIONS	. NO			-) TEST 71-014-056
QU-228 AIRCRAFT INFLIGHT NOISE LEVELS) 10 JAN 75
	¥,	1,1	1	2	LOCATION/CONDITION	NYCON	NITTON	1/8	1/1	
FREQ (HZ)										
31.5	103	101	100	101	103	102	101	102	101	
63	114	102	100	101	101	100	102	101	100	
125	109	112	108	108	108	1.06	106	109	103	
250	103	105	106	102	106	103	104	107	106	
200	96	102	96	96	96	76	46	96	9.0	
1000	91	16	91	68	88	88	87	60	85	
2000	88	46	87	98	85	83	83	85	81	
0004	88	76	87	9 4	83	81	82	83	80	
0009	92	+6	94	82	80	77	7.8	62	92	
16000	82	90	81	19	7.8	73	75	22	7.5	
OVERALL	116	114	111	110	111	109	110	112	109	

3	HUMAN NOISE EXPOSORE	EXPO	SURE) IDENTIFICATION:) OMEGA 3.2
NOISE SOURCE/SUBJECT:	-	OPERATION	ION			-) TEST 71-014-05
OU-228 AIRCRAFT INFLIGHT NOISE LEVELS											Z
						^) PAGE H1
	1/4	1/8	1/0	1/0	LOCATI	ON/CON 1/F	LOCATION/CONDITION 1/E 1/F 1/G	ž	1,1	3	
HAZARD/PROTECTION C-WEIGHTED OVERALL S A-WEIGHTED OVERALL S MAXIMUM PERMISSIBLE NO PROTECTION	TIME	FFF	(OASLC IN (OASLA IN HINUTES) F	N DBC) A N DBA) A FOR ONE		SURE	PER DAY	CAFR	161-35,	JULY	73)
	107	109	112	112	112	112	112	110	117	112	
OASLA	680	90	93	100	100	100	101	86	102	66	
MINIMUM OPL EAR HUFFS	303		:	3	;	3	;	;	13	9	
	679	100	89	89	170	90	170	88	96	90	
V-51R EAR PLUGS											
OASLA*	960	71	960	960	960	960	960	960	807	960	
FLENTS EAR PLUGS	i		i								
DASLAT	960	960	960	080	960	096	807	960	571	62	
H-157 IN-FLIGHT COMMUNICATION	VICATION	UNIT							:		
OASLA*	679	1001	88	170	143	143	92	88	95	91	
COMMUNICATION PREFERRED SPEECH INT PSIL	INTERFERENCE 77	DE LEVEL	EL (PSIL 80	NO B	98	91	06	06	16	69	
ANNOYANCE PERCEIVED NOISE LEVE	LEVEL, TONE		CTED (PNLT I	CORRECTED (PNLT IN PNDB)						
	107	108	113	115	116	116	118	114	121	116	

											OMEGA 3.2
NOISE SOURCE/SUBJECT:	0.0	OPERATIONS	. NO			^) TEST 71-014-056) RUN 02
QU-22B AIRCRAFT											28 APR 76
INFLIGHT NOISE LEVELS											PAGE H1
	1,4	₹	£	Z T	OCATI 1/P	LOCATION/CONDITION 1/P 1/Q 1/R	17.8 17.8	1/5	,		
A-WEIGHTED OVERALL MAXIMUM PERMISSIBLE	SOUND LEVEL TIME (T IN		COASLC IN	08 A		SURE	PER DAY	(AFR	161-35,	JULY 73)	33
NO PROTECTION OASLC	115	114	111	110	111	109	109	112	109		
OASLA	100	104	100	86	100	96	86	101	97		
HINIMUM OPL EAR MUFFS	2	12	30	7	30	24	24	62	20		
	93	92	89	88	89	98	87	89	98		
1	101	120	202	240	202	339	285	202	339		
OASLA*	62	81	82	11	79	77	7.8	80	92		
2011 10 040 2111111	096	807	096	960	096	096	096	960	096		
OASLA*	9.1	82	62	78	80	11	62	81	7.8		
-	807	619	960	960	960	096	960	807	960		
H-157 IN-FLIGHT COMMUNIC	A	UNIT									
OASLA*	101	121	1 90	880	170	285	89	91	888		
COMMUNICATION PREFERRED SPEECH INTE	INTERFERENCE 92	E LEVEL	L (PSIL	L IN 08)	89	6		6	82		
ANNOYANCE PERCEIVED NOISE LEVEL	EVEL, TONE	CORRECTED (PNLT IN PNDB)	TEO (F	WLT IN	PNOB						
-	118	121	111	115	117	11,	115	118	114		